

SUPPLEMENTAL TYPE CERTIFICATE

10048423 REV. 1

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

KAMAN CORPORATION AIRCRAFT WHEEL AND BRAKE, LLC

**1160 CENTER ROAD
AVON OH 44011
UNITED STATES OF AMERICA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: 1. FAA A24CE
2. EASA.IM.A.277

Type Certificate Holder: Beechcraft Corporation

Type: KING AIR 200

Model: 1. 200, 200C, 200CT, 200T
2. B200, B200C, B200CT
3. B200T, B200GT, B200CGT.

Original STC Number: FAA STC SA757GL

Description of Design Change:

Replacement of Main Wheels & Brakes and/or Nose Wheel.
Rev. 1: Transfer from "Parker Hannifin Corporation" to
"Kaman Aviation, Aircraft Wheel and Brake, LLC".

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.
The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 27 July 2023



Dominique RENAUD
Section Manager
Jet Aeroplanes



Associated Technical Documentation:

Installation of Main Wheel and Brakes in accordance with Parker Hannifin Conversion kit 199-125 Rev R, dated September 07, 2010, per Installation Manual, Rev L, dated January 21, 2010, and Installation Drawing 50-90, Rev K, dated January 21, 2010, and/or:-

Installation of Nose Wheel in accordance with Parker Hannifin Conversion kit 199-126 Rev L, dated September 07, 2010, per Installation Manual, Rev G, dated January 21, 2010, and Installation Drawing 50-91, Rev H, dated January 21, 2010.

or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

These modifications are approved for individual or joint installation on specified models that are equipped with landing gear that support 10 inch High Floatation equipment.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

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