

## SUPPLEMENTAL TYPE CERTIFICATE

**10062574 REV. 1**

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### **KAMAN CORPORATION AIRCRAFT WHEEL AND BRAKE, LLC**

**1160 CENTER ROAD  
AVON OH 44011  
UNITED STATES OF AMERICA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** US A25CE

**Type Certificate Holder:** TEXTRON AVIATION INC.

**Type:** 404/406

**Model:** 404

**Original STC Number:** FAA STC SA781GL

#### **Description of Design Change:**

Replace existing Main Wheel and Brakes by installing Parker Hannifin Conversion Kit 199-78.

Rev. 1: Transfer from "Parker Hannifin Aircraft Wheel and Brake Division" to "Kaman Corporation Aircraft Wheel and Brake Division".

#### **EASA Certification Basis:**

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

#### **Associated Technical Documentation:**

Installation Instructions and Installation Drawing 50-54 Rev. C dated 31 May 1984 or later FAA-approved revision.

See Continuation Sheet(s)

**For the European Union Aviation Safety Agency**

**Cologne, Germany, 14 October 2024**



**Marco CAPACCIO**

**Section Manager**

**Small Aircraft, Balloons & Airships**



**Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -