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SERVICE BULLETIN

P/N 30-244 BRAKE ASSEMBLY – INCREASED BLEEDER TORQUE VALUE

1. PLANNING INFORMATION

A. APPLICABILITY

Refer to Table 1. This service bulletin is applicable to the Pilatus Aircraft Ltd models listed using the designated Parker Hannifin Corporation, Aircraft Wheel & Brake Division main brake assembly part numbers:

Table 1 Applicability

Parker Hannifin Part No.	Pilatus Aircraft Ltd Model	Serial Number
30-244	PC-12, PC-12/45, PC-12/47, PC-12/47E	All

B. REASON

To enhance bleeder retention.

C. DESCRIPTION

This service bulletin will provide instruction to the level required for inspection and the return of product to service.

D. COMPLIANCE

Recommended – At earliest convenience or next brake overhaul.

E. APPROVAL

This service bulletin is approved for support of an FAA-TSO-C26 product.

F. MANPOWER

Estimate. 15 minutes per aircraft.

G. TOOLING

Standard tooling:

- Socket, hex head, 0.375 inch
- Torque gage

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H. WEIGHT AND BALANCE

Not affected.

I. REFERENCES

Use the latest issue in effect.

CM30-244.....Component Maintenance Manual with Illustrated Parts List for Main Brake Assembly PHC Part No. 30-244

J. OTHER PUBLICATIONS AFFECTED

CM30-244.....Component Maintenance Manual with Illustrated Parts List for Main Brake Assembly PHC Part No. 30-244

K. PREVIOUS MODIFICATIONS

None.

2. ACCOMPLISHMENT INSTRUCTIONS

Instructions are per brake assembly.

All instruction related to aircraft support and non-related Parker Hannifin, Aircraft Wheel & Brake equipment shall be in accordance with the airframe manufacturer's instructions.

A. INSPECT BLEEDER TORQUE VALUE

SAFETY WARNING:
MAKE SURE THE AIRCRAFT IS SECURE AND STABLE BEFORE
BEGINNING ANY WORK. WORKING AROUND AN AIRCRAFT THAT IS NOT SECURE AND
STABLE CAN CAUSE INJURY OR DEATH.

Refer to Figure 1 and the following procedure.

- (1) Inspection of bleeder torque can be accomplished while equipment is on aircraft.
- (2) Use a torque gage with a .375 inch hex socket attached and check the bleeder torque.
- (3) If the torque value measurement is 18 to 20 in-lb (2.03 to 2.26 N-m):
 - (a) Increase the torque value to 35 to 38 in-lb (4.0 to 4.3 N-m).
- (4) If the torque value measurement is 35 to 38 in-lb (4.0 to 4.3 N-m):
 - (a) The torque value meets the requirements of this service bulletin.

B. LOGBOOK ENTRY

(1) Make the appropriate airframe logbook entry of compliance.

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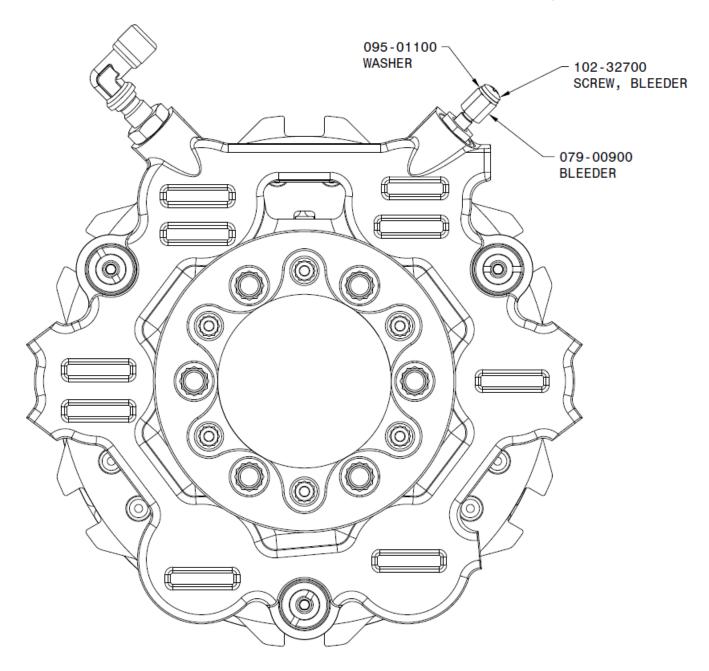


Figure 1 Location of 079-00900 bleeder

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3. MATERIAL INFORMATION

A. MATERIAL-REQUIREMENTS

Not applicable.

B. LIST OF COMPONENTS

Not applicable.

C. INTERCHANGEABILITY

None.

D. PARTS DISPOSITIOIN

Not applicable.

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